TEI-PD170

PISTON ENGINE



MAIDEN FLIGHT WITH ANKA 27 DECEMBER 2018

MAIDEN FLIGHT WITH AKSUNGUR 04 OCTOBER 2021

MAIDEN FLIGHT WITH BAYRAKTAR TB3 27 OCTOBER 2023

FIRST INDIGENOUS TURBODIESEL AVIATION ENGINE OF TÜRKİYE

TECHNICAL SPECIFICATIONS

Displacement

Engine Configuration

Fuel Supply

Air Induction

Maximum Power

Dry Weight

Engine Cooling

Material

Fuel Options

BSFC (@MSL)

Description Electrical Power

Altitude Power(@30,000 ft) Altitude Power(@40,000 ft)

Critical Altitude

Maximum Altitude

Propeller Control Aircraft Configuration

Altitude Power(@20,000 ft)

▶ Engine Control

> FADEC Software

Alternator Control

: 2.1 L

: Straight 4 Cylinder

: Common Rail Diesel Injection

: Two-Stage Turbocharging

: (@2,300 propeller RPM) 172 HP (±2)

: 162 kg

: Water Cooled

: Aluminium Block and Cylinder Head

: JP-8, JET-A1, JP-5, and EN 590

: 207 g/kWh (±3)

: 9 kW (2x4.5 kW)

: 170 HP

: 130 HP : 90 HP

: 20,000+ ft

: 45.000 ft

: ECU Controlled Hydromechanical

: Tractor or Pusher Compatible

: Redundant ECU (FADEC)

: DO-178C DAL-C Certifiable

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ADVANTAGES OF THE TEI-PD170

- High power/weight ratio
- Superior high altitude capability
- Less fuel consumption
- Compact design
- More than 90% localized
- 3,600 hours engine lifetime
- High take-off power
- High altitude power

FLIGHT RECORD 45 HOURS

ALTITUDE RECORD 40.000+ FT

OTHER HIGHLIGHTS

- Turbodiesel aviation engine designed for Medium Altitude Long Endurance (MALE) unmanned aerial vehicles
- Engine is certifiable according to civil aviation engine specifications
- FADEC is compliant with MIL-STD-461F, MIL-STD-704F, MIL-STD-810G
- Reconfigurable engine control system software and hardware (developed by TEI)
- Power increase potential up to 200+ HP
- ▶ Engine is designed and validated according to EASA CS-E